

## Systems with Distributed Mass and Elasticity

### 1. Free vibration of a bending beam

Bending Beam Equation:

$$m\ddot{u} + EIu'''' = 0 \quad (1)$$

Define solution as:

$$u(x,t) = \phi(x)q(t) \quad (2)$$

Substitute Steady state

$$\ddot{u} = -\omega^2 u \quad (3)$$

$$EI\phi'''' - \omega^2 m\phi = 0 \quad (4)$$

$$EI\phi'''' - \beta^4 m\phi = 0 \quad (5)$$

$$\beta^4 = (\omega^2 m / EI) \quad (6)$$

Solution is:

$$\phi(x) = C_1 \sin \beta x + C_2 \cos \beta x + C_3 \sinh \beta x + C_4 \cosh \beta x \quad (7)$$

For a simply supported beam, the boundary conditions are:

$$u(0) = 0, u''(0) = 0, u(L) = 0, u''(L) = 0 \quad (8)$$

Leading to:

$$\sin \beta L = 0 \quad (9)$$

or

$$\beta L = n\pi \quad (10)$$

or:

$$w_n = (n^2 \pi^2 / L^2)(\sqrt{EI / M}) \quad (11)$$

and:

$$\phi_n(x) = C_1 \sin(n\pi x / L) \quad (12)$$

For a cantilever beam:

$$u(0) = 0, u'(0) = 0, u''(L) = 0, u'''(L) = 0 \quad (13)$$

Leading to:

$$1 + \cos \beta L \cosh \beta L = 0 \quad (14)$$

or:

$$w_1 = (3.516 / L^2)(\sqrt{EI / M}), w_2 = (22.03 / L^2)(\sqrt{EI / M}), \quad (15)$$
$$w_3 = (61.70 / L^2)(\sqrt{EI / M}), w_4 = (120.9 / L^2)(\sqrt{EI / M})$$

## 2. Modal Orthogonality

Steady state response of an SDOF system to periodic excitation is obtained at the aid of Fourier Transform and Inverse Fourier Transform.

For a cantilever beam:

$$EI \phi_n'''' = \omega_n^2 m \phi_n \quad (16)$$

Leading to:

$$\int_0^L \phi_r EI \phi_n'''' dx = \omega_n^2 \int_0^L \phi_r m \phi_n dx \quad (17)$$

Leading to:

$$\int_0^L EI \phi_r'' \phi_n'' dx = \omega_n^2 \int_0^L m \phi_r \phi_n dx \quad (18)$$

Replacing everything above but starting with mode r and multiplying by mode n:

$$\int_0^L EI \phi_r'' \phi_n'' dx = \omega_r^2 \int_0^L m \phi_r \phi_n dx \quad (19)$$

Subtracting:

$$(\omega_r^2 - \omega_n^2) \int_0^L m \phi_r \phi_n dx = 0 \quad (20)$$

for:

$$(\omega_r \neq \omega_n) \quad (21)$$

The orthogonality condition becomes:

$$\int_0^L m \phi_r \phi_n dx = 0 \quad (22)$$

or:

$$\int_0^L EI \phi_r'' \phi_n'' dx = 0 \quad (23)$$

## Modal Analysis

Define solution as:

$$u(x,t) = \sum_{r=1}^{\infty} \phi_r(x) q_r(t) \quad (24)$$

Substituting in beam equation:

$$\sum_{r=1}^{\infty} m\phi_r \ddot{q}_r + \sum_{r=1}^{\infty} EI\phi_r'''' q_r = -m\ddot{u}_g \quad (25)$$

Multiply by  $\phi_n$  and integrate over domain  $0-L$ :

$$\int_0^L \phi_n \sum_{r=1}^{\infty} m\phi_r \ddot{q}_r dx + \int_0^L \phi_n \sum_{r=1}^{\infty} EI\phi_r'''' q_r dx = -\int_0^L \phi_n m dx \ddot{u}_g \quad (26)$$

Due to modal orthogonality:

$$\int_0^L m\phi_n^2 dx \ddot{q}_n + \int_0^L \phi_n EI\phi_n'''' q_n dx = -\int_0^L \phi_n m dx \ddot{u}_g \quad (27)$$

or:

$$M_n \ddot{q}_n + K_n q_n = -L_n \ddot{u}_g \quad (28)$$

and as always,

$$K_n = \omega^2 M_n \quad (29)$$

Now we can solve each equation for  $q_n$  independently, and then calculate  $u$ :

or:

$$\ddot{q}_n + 2\zeta_n \omega_n \dot{q}_n + \omega_n^2 q_n = -(L_n / M_n) \ddot{u}_g \quad (30)$$

and,

$$u(x,t) = \sum_{r=1}^{\infty} \phi_r(x) q_r(t) \quad (31)$$